

SUBJECT: CHANGE PROPOSALS LAC-160 AND 161

- 1. BOTH PROPOSALS ARE APPROVED BY THE FOG SPO WITH THE FOLLOWING CONSIDERATIONS:
- A. THE FLIGHT TEST EVALUATION OF THE COMPETING AUTOPILOT SYSTEM IS TO BE ACCOMPLISHED WITHOUT THE IMPROVED NAVIGATION CAPABILITY (LAC-160) INSTALLED. THIS IS IN ORDER TO AVOID POSSIBLE INFLUENCE OF NAVIGATION CONVENIENCE ON AUTOPILOT SUBJECTIVE EVALUATION.
- B. A MINIMUM OF ONE AFSC AND ONE SAC PILOT WILL BE REQUIRED TO FLY THE TEST INSTALLATIONS FOR EVALUATIONS. SUGGEST THIS EFFORT BE DONE OUT AT EDWARDS.
- C. THE NAVIGATION IMPROVEMENT SHOULD BE INSTALLED AFTER THE AUTOPILOT EVALUATION HAS BEEN COMPLETED AND FLOWN BY TWO SAC PILOTS ON SIMULATED MISSION PROFILES.

-END OF MSG-

SECRET

GROUP 1 Excluded from automatic downgrading and declassification 2

LOCKHEED AIRCRAFT CORPORA	ENGINEER CHANGE	PROPOSAL [10170096 - , Д. С	-161	
5 NOVEMBER 1963	AFFECTS:	WSP	o [x]	PRO	JECT X	
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SP-F-1A

REASON FOR PROPOSAL:

To evaluate and compare performance and reliability of two combination automatic flight control (autopilot) and flight reference systems with improved system functions, improved control features, and improved pilot operation.

NATURE OF PROPOSAL:

Part A

Install, in breadboard form, a prototype LSI proposed AFCS and flight reference system.

The AFCS will be similar to one currently used in the Gyrodyne and Q2C Drone systems using proven electronics circuits and will utilize existing ships AFCS components where practical.

The flight reference system will be LSI 2 gyro platform type AF/A24G-1A.

An improved course indicator will be installed in place of the ID-250.

Part B

Upon completion of flight test of Part A, the LSI system will be removed and a prototype Minneapolis Honeywell AFC system will be installed in breadboard manner along with a Bendix 2 gyro platform flight reference system.

The AFCS (M-H) will adapt and modify an autopilot electronics system originally designed and packaged for the F-104J aircraft. It will include an air data computer which will be a sub-system designed and packaged for another program and modified to provide true air speed information to the new navigation system (see ECP LAC-160).

The flight reference system will be an A/A24G-5 including 2 gyro Bendix platform, AN/AJN/3A servo amplifier, AN/AJN-3 compass control panel, flux valve, and rate switch.

An improved course indicator will be installed in place of the ID-250.

